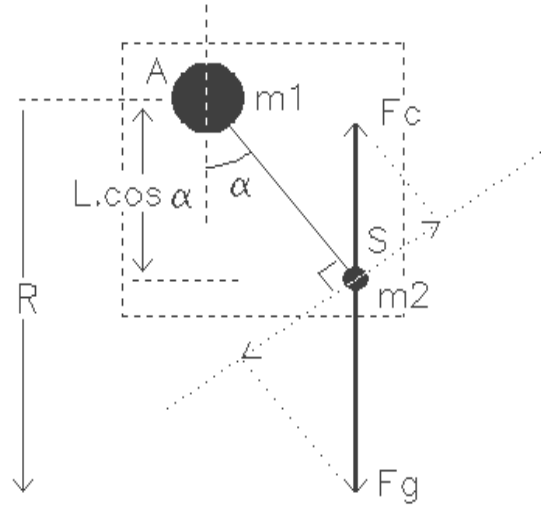


Solution of question 2.

a_1 - Since $m_2 \ll m_1$, the Atlantis travels around the earth with a constant speed. The motion of the satellite is composed of the circular motion of the Atlantis about the earth and (possibly) a circular motion of the satellite about the Atlantis.

For m_1 we have:

$$m_1 \cdot \Omega^2 \cdot R = \frac{G \cdot m_1 \cdot m_a}{R^2} \rightarrow \Omega^2 = \frac{G \cdot m_a}{R^3}$$



For m_2 we have:

$$m_2 \cdot L \cdot \ddot{\alpha} = -(F_g - F_c) \cdot \sin(\alpha) = -\left(\frac{G \cdot m_2 \cdot m_a}{(R - L \cdot \cos(\alpha))^2} - m_2 \cdot \Omega^2 \cdot (R - L \cdot \cos(\alpha)) \right) \cdot \sin(\alpha)$$

Using the approximation:

$$\frac{1}{(R - L \cdot \cos(\alpha))^2} \approx \frac{1}{R^2} + \frac{2 \cdot L \cdot \cos(\alpha)}{R^3}$$

and equation (1), one finds:

$$L \cdot \ddot{\alpha} = -\left(\frac{G \cdot m_a}{R^2} + \frac{2 \cdot G \cdot m_a}{R^3} \cdot L \cdot \cos(\alpha) - \frac{G \cdot m_a}{R^3} \cdot R + \frac{G \cdot m_a}{R^3} \cdot L \cdot \cos(\alpha) \right) \cdot \sin(\alpha)$$

so:

$$\ddot{\alpha} + 3 \cdot \Omega^2 \cdot \sin(\alpha) \cdot \cos(\alpha) = 0 \quad (2)$$

$$\begin{aligned} \text{If } \alpha \text{ is constant: } \ddot{\alpha} = 0 \quad & \rightarrow \sin(\alpha) = 0 \quad \rightarrow \alpha = 0; \quad \alpha = \pi \\ & \rightarrow \cos(\alpha) = 0 \quad \rightarrow \alpha = \pi/2; \quad \alpha = 3\pi/2 \end{aligned}$$

a₂ - The situation is stable if the moment $M = m_2 L \ddot{\alpha} L = m_2 L^2 \ddot{\alpha}$ changes sign in a manner opposed to that in which the sign of $\alpha - \alpha_0$ changes:

sign($\alpha - \alpha_0$)	-	+	-	+	-	+	-	+	-	+
α	0	$\pi/2$	π	$3\pi/2$	2π					
sign(M)	+	-	-	+	+	-	-	+	+	-
α	0	$\pi/2$	π	$3\pi/2$	2π					

The equilibrium about the angles 0 en π is thus stable, whereas that around $\pi/2$ and $3\pi/2$ is unstable.

b - For small values of α equation (2) becomes:

$$\ddot{\alpha} + 3\Omega^2 \alpha = 0$$

This is the equation of a simple harmonic motion.

The square of the angular frequency is:

$$\omega^2 = 3\Omega^2$$

so:

$$\omega = \Omega \sqrt{3} \quad \rightarrow \quad T_1 = \frac{2\pi}{\omega} = \frac{1}{3} \sqrt{3} \left(\frac{2\pi}{\Omega} \right) \approx 0,58 T_0$$

c₁ - According to Lenz's law, there will be a current from the satellite (S) towards the shuttle (A).

c₂ - For the total energy of the system we have:

$$U = U_{kin} + U_{pot} = \frac{1}{2} m \Omega^2 R^2 - \frac{G m m_a}{R} = -\frac{1}{2} \frac{G m m_a}{R}$$

A small change in the radius of the orbit corresponds to a change in the energy of:

$$\Delta U = \frac{1}{2} \frac{G m m_a}{R^2} \Delta R = \frac{1}{2} m \Omega^2 R \Delta R$$

In the situation under c₁ energy is absorbed from the system as a consequence of which the radius of the orbit will decrease.

Is a current source inside the shuttle included in the circuit, which maintains a net current in the opposite direction, energy is absorbed by the system as a consequence of which the radius of the orbit will increase.

From the assumptions in c₂ we have:

$$\Delta U = F_i v t = B I L \Omega R t = \frac{1}{2} m \Omega^2 R \Delta R \quad \rightarrow \quad t = \frac{1}{2} \frac{m \Omega \Delta R}{B I L}$$

Numerical application gives for the time: $t \approx 5,8 \cdot 10^3$ s; which is about the period of the system.

Marking breakdown:

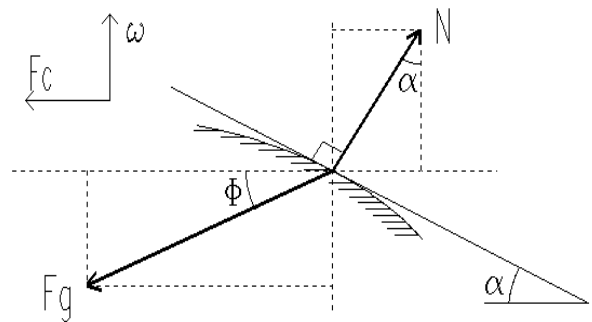
a ₁	: 1
a ₂	: 1
b - Atlantis in uniform circular motion	: 0,5
- calculation of the period Ω	: 0,5
- equation of motion of the satellite	: 2,5
- equation of motion for small angles	: 0,5
- period of oscillations	: 1
c ₁ -	: 1
c ₂ - calculation of the time the current has to be maintained	: 1,5
- increase or decrease of the radius of the orbit	: 0,5

Solution of question 3.

a - 1st method

For equilibrium we have $F_c = F_g + N$
 where N is normal to the surface.

Resolving into horizontal and vertical components, we find:



$$F_g \cdot \cos(\phi) = F_c + N \cdot \sin(\alpha)$$

$$F_g \cdot \sin(\phi) = N \cdot \cos(\alpha) \quad \rightarrow \quad F_g \cdot \cos(\phi) = F_c + F_g \cdot \sin(\phi) \cdot \text{tg}(\alpha)$$

From:

$$F_g = \frac{G \cdot M}{r^2}, \quad F_c = \omega^2 \cdot r, \quad x = r \cdot \cos(\phi), \quad y = r \cdot \sin(\phi) \quad \text{en} \quad \text{tg}(\alpha) = \frac{dy}{dx}$$

we find:

$$y \cdot dy + \left(1 - \frac{\omega^2 \cdot r^3}{G \cdot M} \right) \cdot x \cdot dx = 0$$

where:

$$\frac{\omega^2 \cdot r^3}{G \cdot M} \approx 7 \cdot 10^{-4}$$

This means that, although r depends on x and y, the change in the factor in front of xdx is so slight that we can take it to be constant. The solution of Eq. (1) is then an ellipse: